

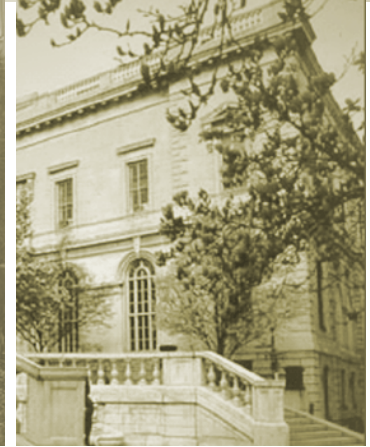
Summary of Capabilities in Spacecraft and Mission Design

*Presentation to
Steering Group
Planetary Decadal Survey
July 7, 2009*

*Dr. John C. Sommerer
Director, S&T and
Acting Space Department Head
john.sommerer@jhuapl.edu
240 228 6242*

APL
The Johns Hopkins University
APPLIED PHYSICS LABORATORY

Divisions & Neighbors of The Johns Hopkins University



School of Arts & Sciences

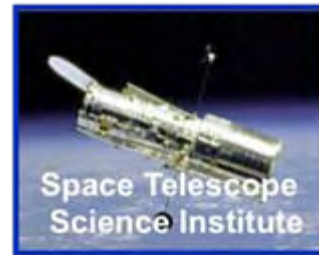
Whiting School of Engineering

School of Professional Studies in Business & Education

**School of Medicine
School of Hygiene & Public Health**

School of Nursing

Applied Physics Laboratory



Nitze School of Advanced International Studies

Peabody Institute

Profile of the Applied Physics Laboratory



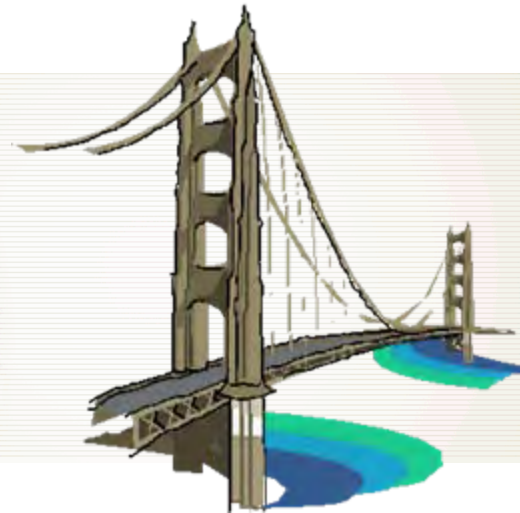
- **Not-for-profit university research & development laboratory**
- **Division of the Johns Hopkins University founded in 1942**
- **On-site graduate engineering program in 8 degree fields**
- **Staffing: 4,500 employees (69% scientists & engineers)**
- **Annual revenue ~ \$ 950M**

What is Our Typical Role?

Government



Industry



Concept Development

- Problem Definition
- System Concept
- Critical Technology Identification
- Demonstration, Validation, Prototyping

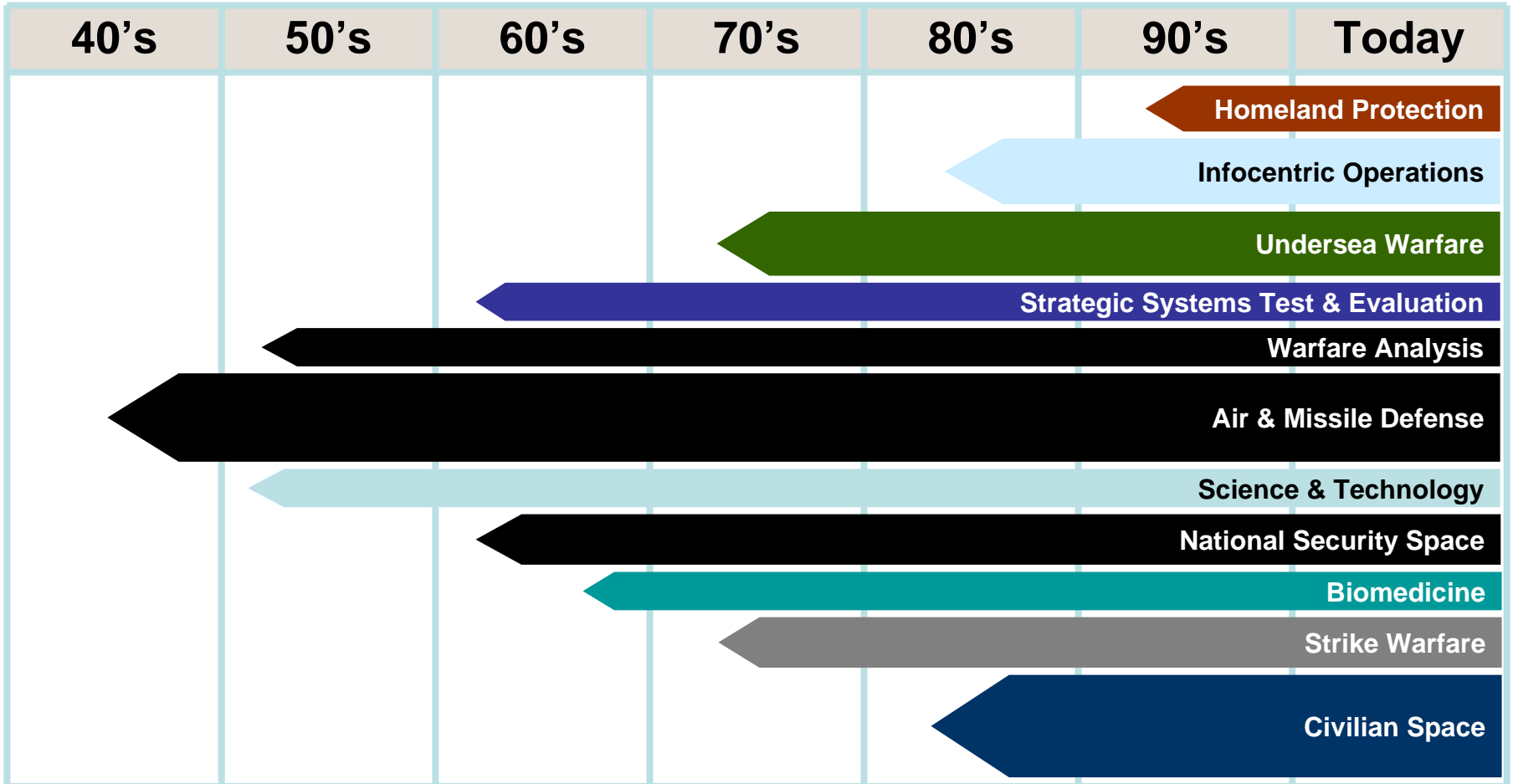
Requirements Definition

- System Design Requirements
- Design Development with Government & Industry
- Technical Evaluation
- Coordination of Integration Testing

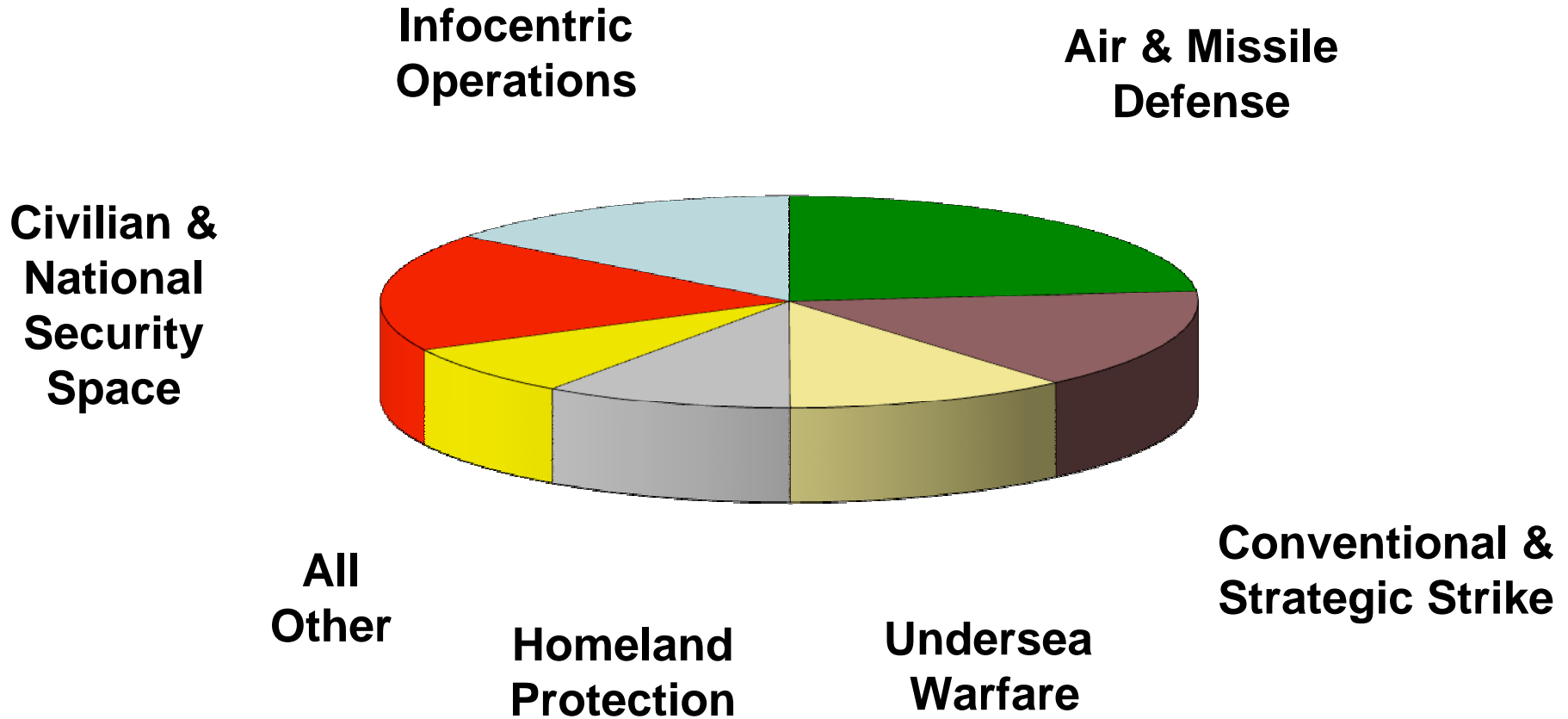
Production & Deployment

- Transition of Prototype Design
- Follow-on Review and Requirements

Business Area History

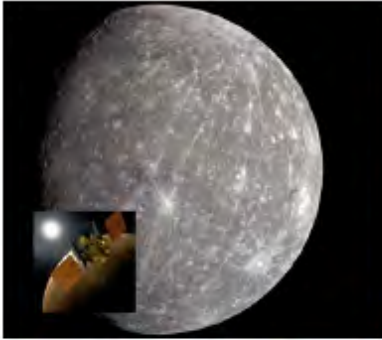


Principal Areas of Work – GFY2009



Example Programs

From the 2008 Annual Report



Space Department at a Glance

- **Department hosts two APL Business Areas:**
 - **Civilian Space**
 - **National Security Space**
- **Three functional branches support these two business areas:**
 - **Software Engineering**
 - **Hardware Engineering**
 - **Science Research and Analysis**
- **Vital Statistics**
 - **Approx. 600 staff, plus 50 on-site contractors**
 - **Approximately 150 space scientists**
 - **\$150-200M Funding**
 - **Sponsors: NASA, DoD, National Science Foundation, National Oceanic and Atmospheric Association (NOAA)**

Current Operations

- **We are currently operating five spacecraft:**
 - ***MESSENGER*** – Launched 2004, First spacecraft to orbit Mercury, 1st & 2nd Flyby 2008, Final Orbit 2011
 - ***New Horizons*** – Launched 2006, First Mission to Pluto, Jupiter Flyby Feb 28 2007; Pluto Flyby 2015
 - ***TIMED*** – Launched 2001, Earth Orbit, study Earth's upper atmosphere
 - ***STEREO*** – Launched Fall 2006, Twin spacecraft around the Sun to study coronal mass ejections
- **Example Active Instruments**
 - **LECP** on Voyager
 - **CRISM** on MRO
 - **Mini-RF** on ISRO Chandrayaan-1 (Mini-RF on LRO)
 - **MIMI** on Cassini
 - **PEPSSI** on New Horizons
 - **GUVI/SSUSI** on TIMED/DMSP
- **Current Spacecraft Engineering**
 - **Radiation Belt Storm Probes**, twin spacecraft to investigate Van Allen Belts, recently confirmed for Phase C

End-To-End Capability Space Science & Exploration

- Publish Scientific Advances
- Work with National Academy & NASA in developing roadmaps for future

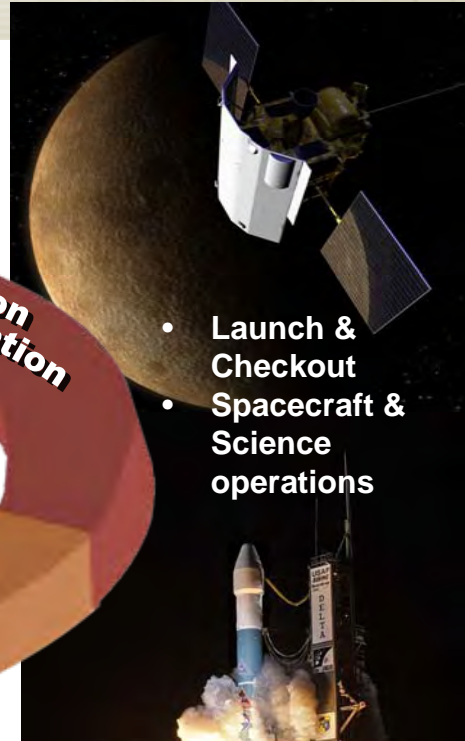
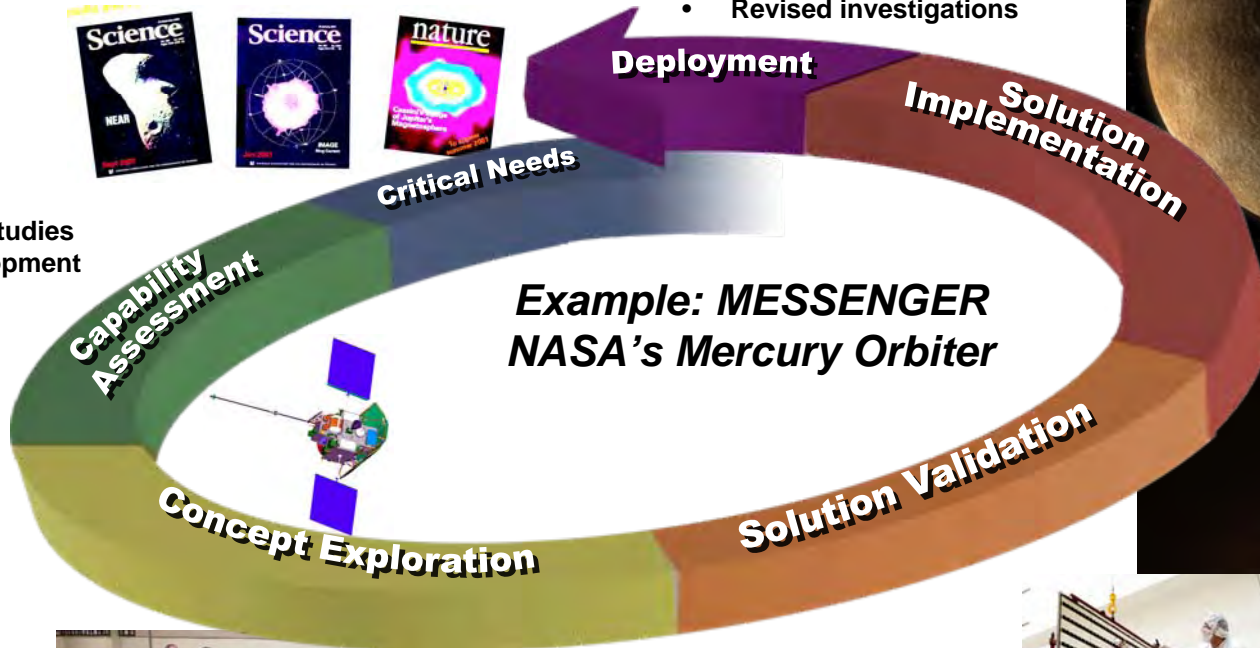
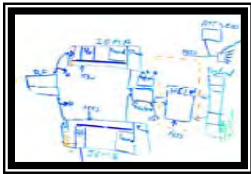
Phase E:

- Operations & Sustainment
- Data Acquisition and Analysis
- Publication of results
- Revised investigations



Pre Phase A:

- Mission Concept Studies
- Technology Development
- Mission Proposals



- Launch & Checkout
- Spacecraft & Science operations

Phase A:

- Mission Concept Development
- Preliminary Mission Analysis



Phase C

- detailed design

Phase D

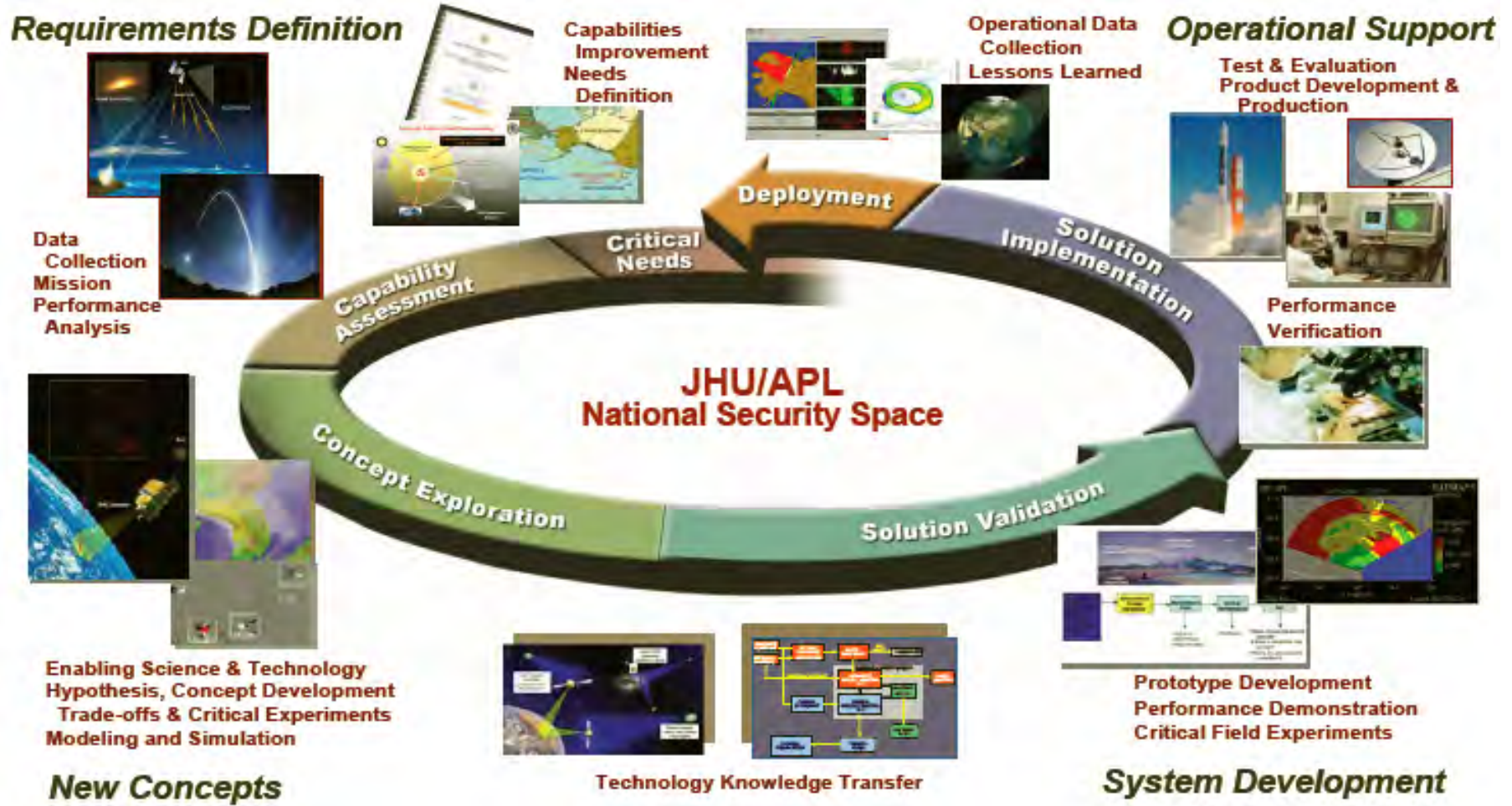
- fabrication, assembly, integration & test
- Hardware Implementation



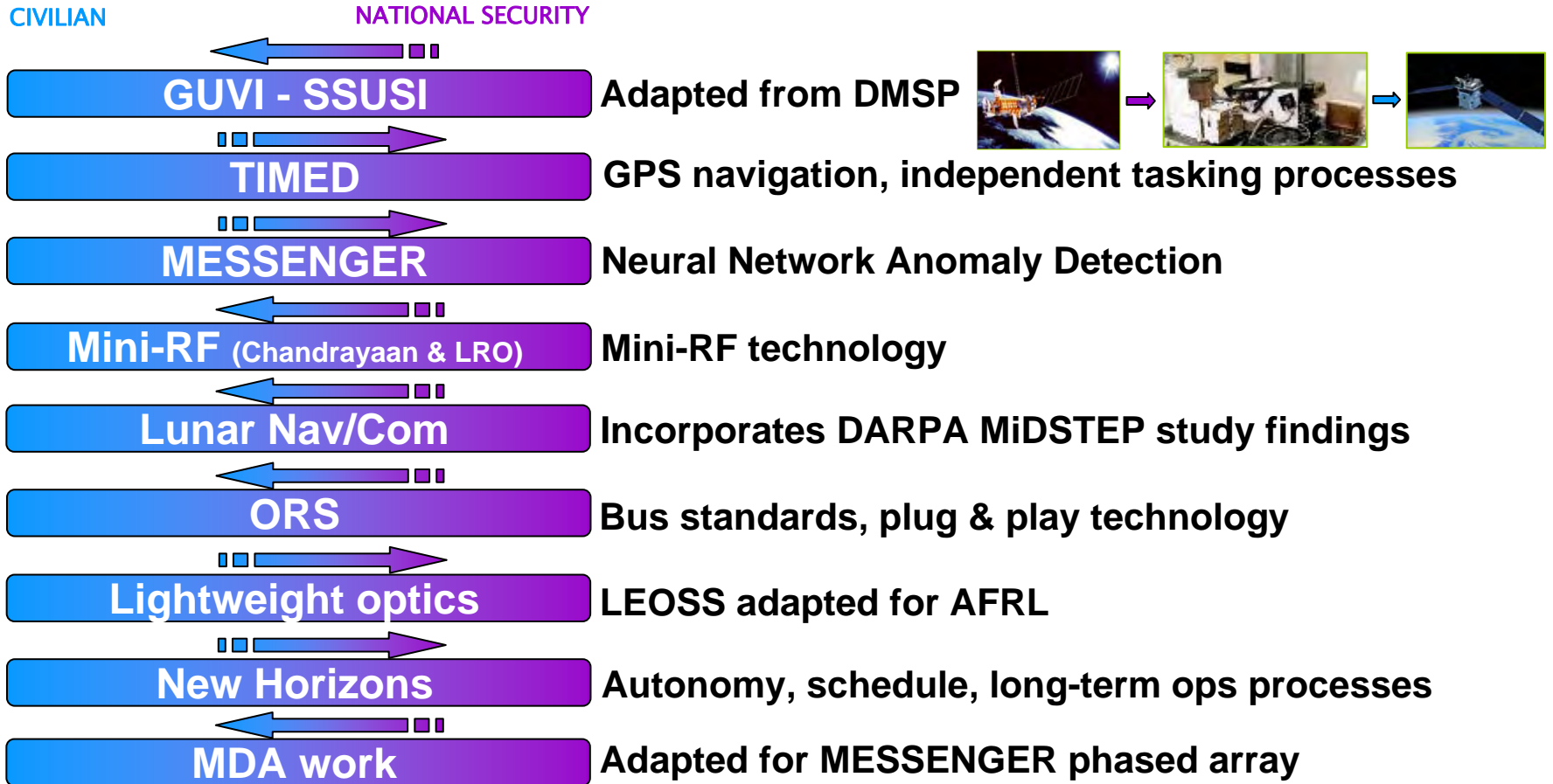
Phase B:

- Definition
- Preliminary Design

End-To-End Capability Space-based Support of National Security



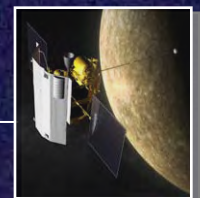
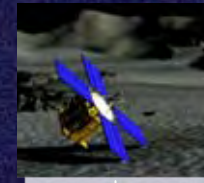
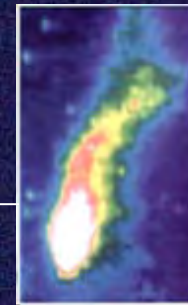
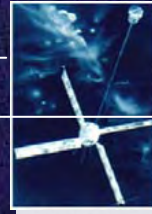
Civilian & National Security Space Synergy



Other examples include Mars ATD low power converter for NSS programs, an NRO instrument adapted for RBSP, CS thermal switch used by USNA MidStar, CS FLAPS IRAD used on USAFA Falconsat

A Tradition of Firsts in Space

- 1958 Satellite Navigation System
- 1961 Nuclear-powered spacecraft
- 1963 Gravity gradient stabilization
- 1967 Color picture of the full Earth
- 1972 Drag-compensated satellite
- 1975 Pulsed plasma thrusters
- 1982 Autonomous satellite navigation with GPS
- 1984 Artificial Comet
- 1986 Intercept of a thrusting target in space
- 1988 Autonomous target acquisition and Track
- 1996 Near Earth Asteroid Rendezvous (NEAR)
- 1996 First Hyperspectral Imager in Space (MSX)
- 1996 Invention of Polymer Battery
- 2001 Landing on an Asteroid (NEAR)
- 2002 Re-Configurable Self-Repairing Processor (on FEDSAT)
- 2003 First orbital Mercury exploration mission launched (MESSENGER)
- 2004 First mission to Pluto launched (New Horizons)



The Hopkins Ultraviolet Background Explorer

HUBE

A Spectroscopic and Imaging Survey of the Sky

Small Explorer Mission of Opportunity

Submitted in response to NASA AD NIN062DA000

January 15, 2006

EAGLES

A Space Weather Science Mission for the Price of an Instrument

US: Normal Incidence Extreme Ultraviolet Spectrograph

Investigation of Solar Dynamics

Invited proposal in response to AD NIN062DA000

Investigator: Joseph Davila, NASA/GSFC

1, 2006

Quest

Quest for European Seas and Tides

MIST

Mars Imager/Imaging Spectrograph

Volume 1 Investigation and Technical Plan

CORNELL UNIVERSITY

RECON

Rendezvous with a Comet Nucleus

GENCORP AEROSPACE

Strata:

Ground Penetrating Radar for the Mars Science Laboratory

Solar Climate Explorer

Determining the Causes of Global Solar Irradiance Variability and Its Impact on Earth Climate

TARANIS XGRE

Exploration of new acceleration and atmosphere-to-magnetosphere coupling processes with high-energy X-rays, Gamma rays and Relativistic Electrons

Response to AD NIN072DA000

January 15, 2006

Treatment Luminous Event (TLE) series are thought to be related by strong current lightning events.

INTERSTELLAR PATHFINDER

A MISSION TO THE INNER EDGE OF THE

Response to NASA AO NIN072DA0030

January 15, 2006

PIETRO

Plasma Ions and Electrons, energetic particles, and magnetic fields for the solar-terrestrial Relations Observatory

Birkeland: An Explorer to Reveal the Invisible Electrodynamics of Active Auroral Forms

PIETRO

Plasma Ions and Electrons, energetic particles, and magnetic fields for the solar-terrestrial Relations Observatory

RB-SPICE

RB-SPICE Radiation Belt Science of Protons, Ion Composition, and Electrons

Radiation Belt Storm Probes

Response to NASA AD NIN062DA0030

November 22, 2005

Farside

Illuminating Early Planetary History

February 13, 2004

A proposal to NASA for a South Pole-Aitken Basin

ALADDIN

Phoenix-Deliver Sample Return Mission

An Innovative Paradigm for Sample Return

A Step-Over Proposal to NASA

August 1996

Advanced ESSP

Global Ice Radar-AEGIR

MOMA

Mars Organic Molecule Analyzer

A Mars Scout Mission of Opportunity investigation in response to NASA AD NIN062DA0030

HELIX

Heliospheric Links Explorer

Jupiter Polar Orbiter:

Exploring the Interior and Dynamical Evolution of a Gas Giant

ABYSS

Altimetric Bathymetry from Surface Slopes

Site Visit

AMPERE

Active Magnetosphere and Planetary Electrodynamics Response Experiment

A Geospace-Related Mission of Opportunity investigation in response to NASA AD NIN062DA0030

November 22, 2005

Tycho

Comet Sampling and Impacts

Response to NASA AD NIN062DA0010

HADLEY

Response to NASA AD NIN062DA0010

First Investigation of Global Wind

Strofio:

Exospheric Sampling of Mercury's Surface Composition

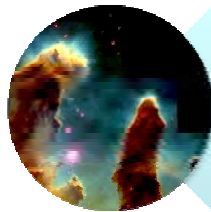
A Discovery Mission of Opportunity investigation in response to NASA AD NIN062DA0010

April 5, 2006

Study & Proposal Examples

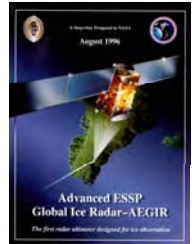


Planetary
Titan Saturn System Mission



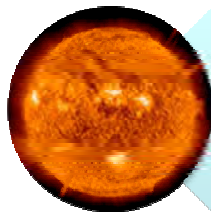
Astrophysics
Interstellar Pathfinder

→ IBEX
SwRI



Earth Science
AEGIR

→ CryoSat
ESA



Heliophysics
HELIX

→ STEREO

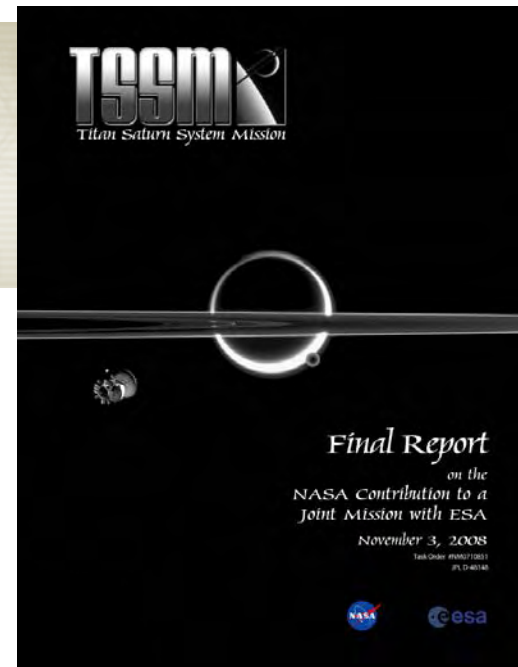
Titan Saturn System Mission

An APL Led Outer Planet Flagship Mission Study

Goal A: Explore Titan, an Earth-like System

Goal B: Examine Titan's Organic Inventory – A Path to Prebiological Molecules

Goal C: Explore Enceladus and Saturn's Magnetosphere – Clues to Titan's Origin and Evolution



Science Team

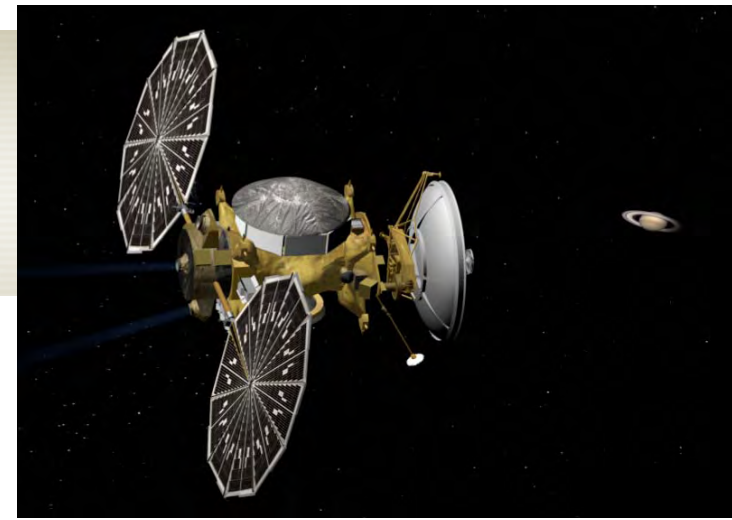
Jonathan Lunine, Co-Chair (University of Arizona)
Jean-Pierre Lebreton, Co-Chair (ESA/ESTEC)
Athéna Coustenis, European Lead Scientist (Observatoire de Paris-Meudon)
Dennis Matson, NASA Study Scientist (JPL)
Lorenzo Bruzzone, University of Trento
Maria-Teresa Capria (Istituto di Astrofisica Spaziale, Rome)
Julie Castillo-Rogez (JPL)
Andrew Coates (Mullard Space Science Laboratory, Dorking)
Michele K. Dougherty (Imperial College London)
Candice Hansen (JPL)
Andy Ingersoll (Caltech)
Ralf Jaumann (DLR Institute of Planetary Research, Berlin)
William Kurth (University of Iowa)
Luisa M. Lara (Instituto de Astrofísica de Andalucía, Granada)
Rosaly Lopes (JPL)
Ralf Lorenz (JHU/APL)

Chris McKay (ARC)
Ingo Muller-Wodarg (Imperial College London)
Olga Prieto-Ballesteros (Laboratorio de Geología Planetaria, Madrid)
François Raulin (LISA Universités Paris 12 & Paris 7)
Amy Simon-Miller (GSFC)
Ed Sittler (GSFC)
Jason Soderblom (University of Arizona)
Frank Sohl (DLR Institute of Planetary Research, Berlin)
Christophe Sotin (JPL)
Dave Stevenson (Caltech)
Ellen Stofan (Proxemy Research)
Gabriel Tobie (Université de Nantes)
Tetsuya Tokano (Universität zu Köln)
Paolo Tortora (Università di Bologna)
Elizabeth Turtle (JHU/APL)
Hunter Waite (SwRI)

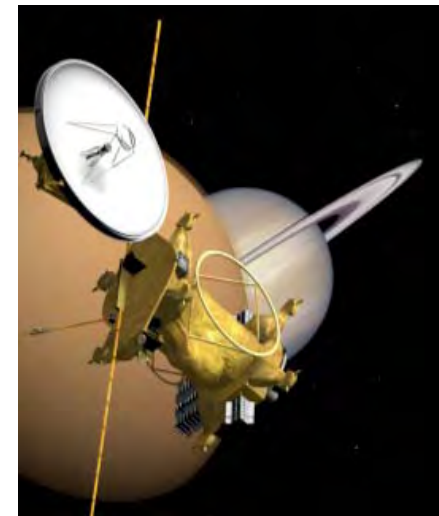
Titan Saturn System Mission

An APL Led Outer Planet Flagship Mission Study

- **2020 Launch, Gravity Assist, SEP**
 - 6203 kg total launch mass, Atlas V 551
- **9 yrs to Saturn, SEP release after 5 yrs**
- **Montgolfiere release 1st flyby, Lander 2nd**
- **4 yr prime, 2 yr Saturn tour, 20 mo Titan orbit**
- **Orbiter (NASA)**
 - 4m high gain antenna 35W Ka-band amplifier for high data downlink
 - 5 ASRG provide 540 W at EOM (MMRTG compatible)
 - Solar Electric Propulsion, 3 NEXT thrusters
 - Two 7.5kW Orion CEV Ultraflex solar arrays
 - 165 kg payload
- **Montgolfiere (ESA)**
 - US MMRTG buoyancy, 10km cruise alt, 6 mo mission
 - 600kg launch mass, telecom with orbiter via 0.5m HGA
- **Lander (ESA)**
 - 9 hour mission, battery operated, X-band omni, 190 kg launch mass

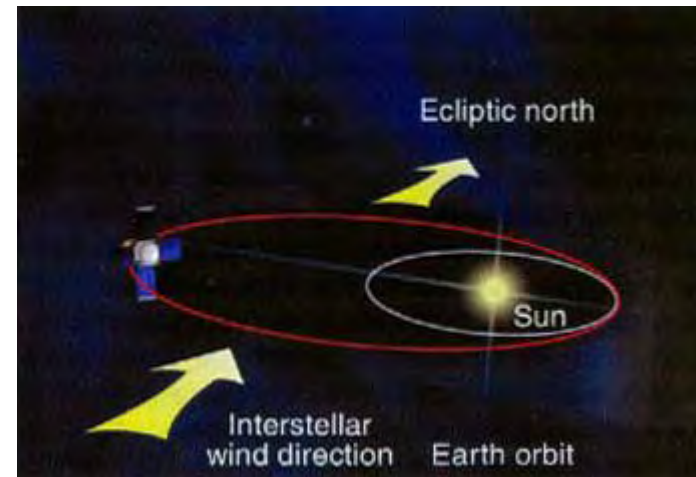


\$2.5B (FY07)
\$1B ESA



Interstellar Pathfinder

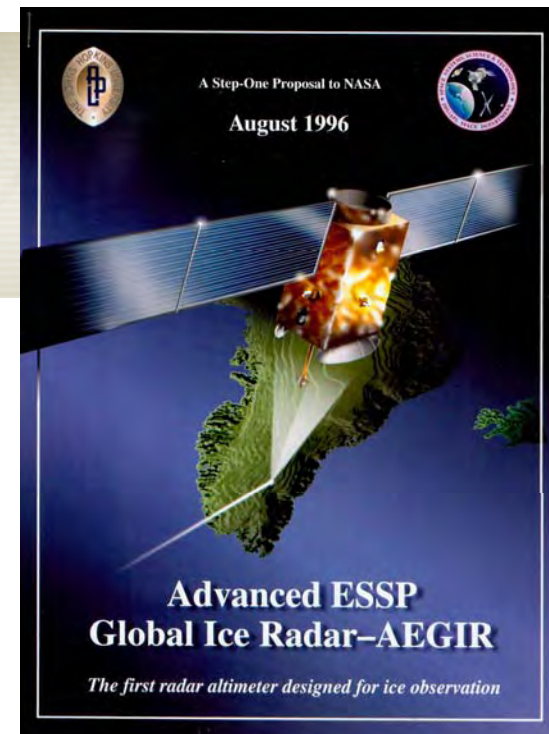
- “...a mission to the inner edge of the interstellar medium.”
 - Composition of local interstellar matter
 - Global images of heliosphere boundary region at 100 to 150 AU
- Submitted 1998 and 2001
- PI George Gloeckler, Univ of Michigan
- ‘98 Spacecraft (Orbital Sciences Corp)
 - July 2006 Launch, 2.3 yr mission
 - 854kg launch mass
 - Orbit 1 to 3 AU, 0° inclination
 - Sun oriented, spin stabilized
 - 1.5m fixed HGA, 10 m² solar array
- ‘98 Payload (37kg, 29W)
 - Ion Composition Spectrometer, Energetic Neutral Atom Imager, Helium Detector, Solar UV Monitor, Solar Wind Monitor



AEGRIR

Advanced ESSP Global Ice Radar

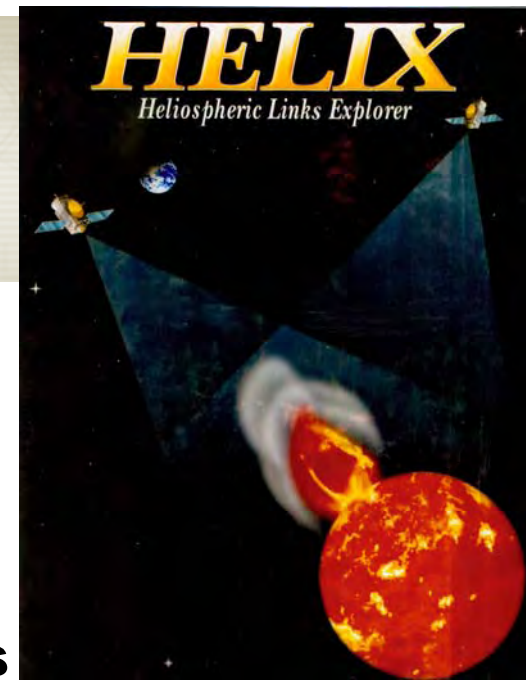
- First radar altimeter designed to observe global ice
- Submitted 1996
- PI Keith Raney, JHU/APL
- Uses APL's patented Delay/Doppler technology at Ku-Band
- Builds on APL experience with Geosat, Seasat, and TOPEX
- Spacecraft
 - Sun Synchronous
 - 880 km altitude
 - 300 W
 - 2001 launch, 5 yr mission
- International Team Members take concept to ESA's CryoSat proposal



HELIX

Heliospheric Links Explorer

- “... provide stereoscopic images of solar mass ejections from their origins on the Sun to their arrival at Earth.”
 - Track eruptions from Sun to Earth
 - Test magnetic helicity conservation
 - Develop geomagnetic storm forecast techniques
- Submitted 1995
- 2 identical spacecraft
 - Heliocentric orbits at 1 AU
 - HELIX-1 remains within 0.2 AU of Earth
 - HELIX-2 drifts away from Earth at 30° per year
- Payload
 - Interplanetary Disturbance Imager
 - Solar Ejection Telescope
 - Plasma and Magnetic Field Sensors

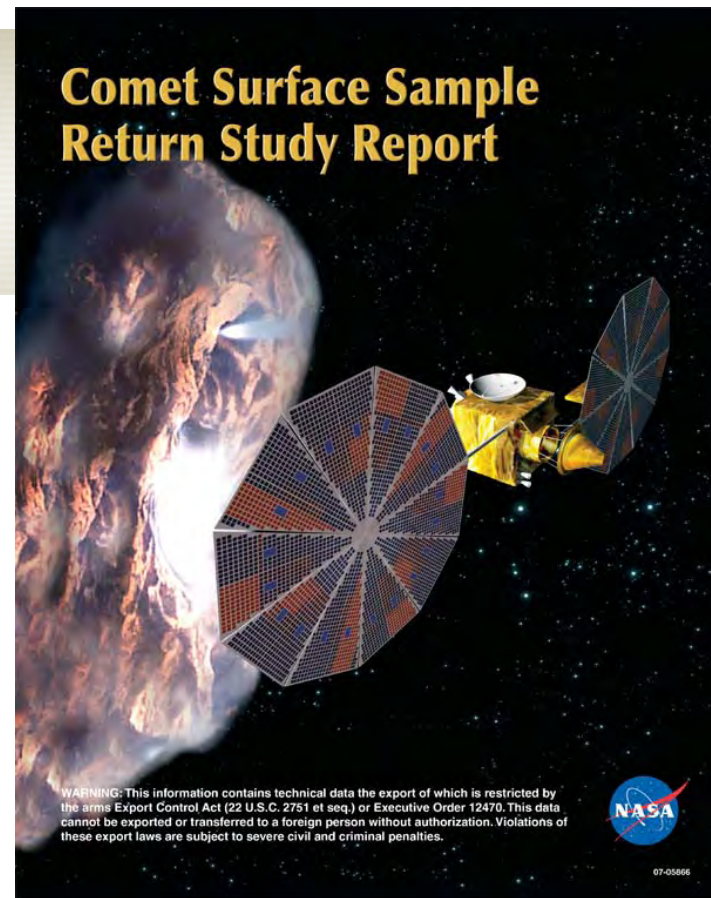


Comet Surface Sample Return

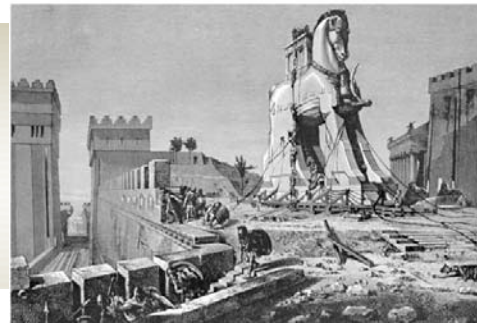
- APL led study shows that a CSSR mission can fit a \$820M (FY07) cost cap
- SDT of community comet experts set reqmts
- Acquire & return 500cc (or more) sample from surface of any comet nucleus
- Return material from a depth of at least 10cm
- Do not allow aqueous alteration
- Capture evolved gasses, maintaining elemental and molecular integrity

Science Definition Team

- Mike A'Hearn (UMD): Co-chair
- Hal Weaver (JHU/APL): Co-chair
- Mike Combi (University of Michigan)
- Yan Fernández (University of Central Florida)
- Will Grundy (Lowell Observatory)
- Martha Hanner (University of Massachusetts)
- Casey Lisse (JHU/APL)
- Karen Meech (University of Hawaii)
- Joe Veverka (Cornell University)
- Paul Weissman (JPL)
- Mike Zolensky (NASA/JSC)



ROSES DSMCE ASRG Study



**ILION:
AN ASRG-
ENABLED
TROJAN
ASTEROID
MISSION**



Science Objectives: Ilion addresses critical questions about Trojan asteroids:

1. Did the Trojan asteroids originate near Jupiter's orbit, or further out in the solar system?
2. How much ice and organics are present on Trojan asteroids?
3. How do the geological processes which have occurred on the Trojans compare to those that have affected other small bodies?
4. What is the relationship between Trojan asteroids and comets, KBOs, and outer planet satellites?

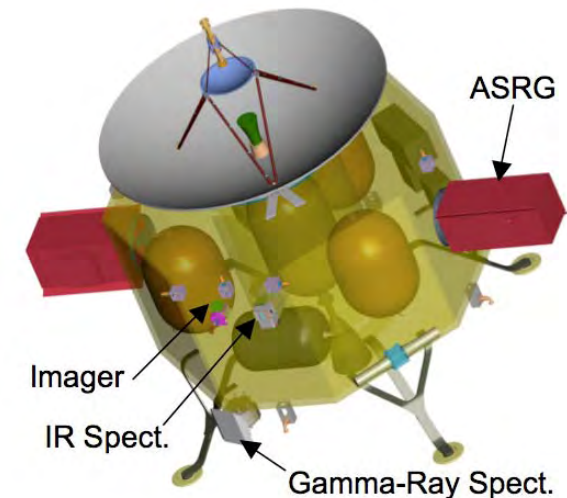
Mission Schedule:

Launch date depends on final target selection

Project Start	August 2009
Launch	September 2013
Jupiter Flyby	May 2016
Trojan Orbit Insertion	August 2023
Land on Trojan	May 2024
End of Mission	September 2024

	Instrument	Heritage/Analog
Remote Sensing	Wide/Narrow Field Imager	MESSENGER
	IR Spectrometer	MESSENGER
	Laser altimeter	NEAR
	Neutron Spectrometer	MESSENGER
	Gamma-Ray Spectrometer	MESSENGER
Surface Sensing	APXS (LAMS? LIBS?)	MER
	Thermal Conductivity	Rosetta et al.
	Electrical Dissipation	

Strawman Payload



NASA SmallSat Study Summary

Study Lead: George Cancro, George.Cancro@jhuapl.edu; 443-778-3476

“The Explorers Program is initiating a study with the Applied Physics Laboratory with the purpose of invigorating the Small Explorer program by introducing innovating approaches to lower cost and improve performance of spacecraft and instrument platforms suitable for future small satellite (SmallSat) missions.”

TASK 1 – Component and Spacecraft Catalog:

- Developed SmallSat Catalog of 300+ different space components, 17 different spacecraft platforms, from 100+ different companies and organizations.

TASK 2 – Reliability, Parts, and Redundancy:

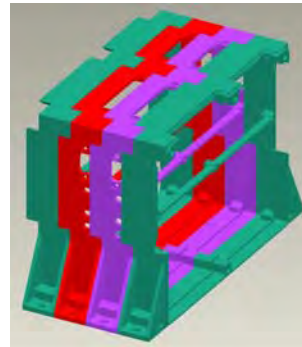
- Augmented failure databases (1962-1995) by adding recent data (1995-2007)
- Based on results as categorized in the database, leads us to conclusion design issues have greater impact than parts and redundancy.

TASK 3 – DoD Dual-Use Technologies:

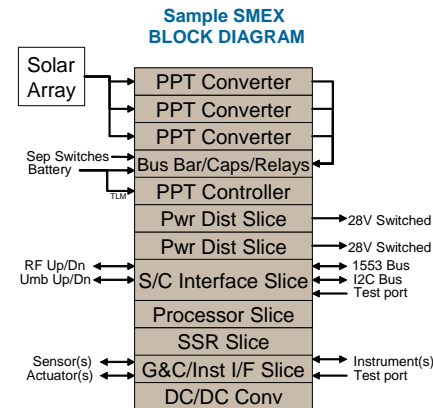
- Surveyed multiple DoD programs resulting in several lessons learned and a RFI to launch providers to gauge near future secondary capabilities.

TASK 4 – NASA SmallSat Standards:

- Developed modular strategies for single box modular avionics (CDH+G&C+PwrDist+SolarArray control for ~8kg and ~30 W) and modular plug-n-play software applications.
- NASA RBSP, ILN, and Solar Probe Plus missions using elements of SmallSat Task 4 concepts**

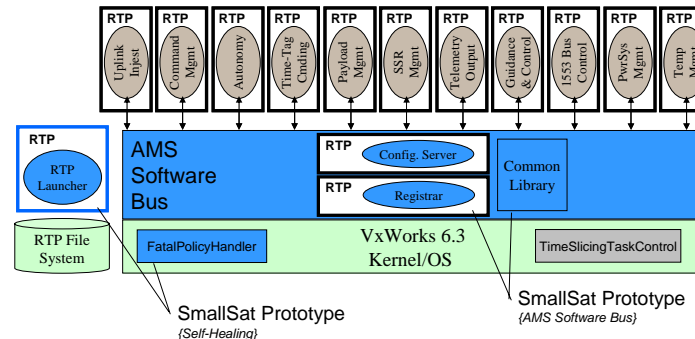


Modular Stacking Slices



SCIF Board
(H/W framer, RAM-based S/W I/F; Transitioned to RBSP)

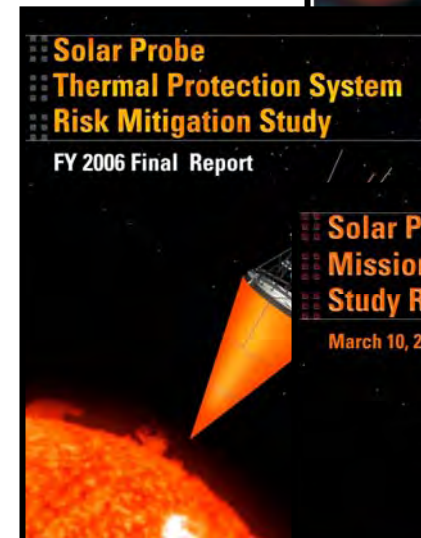
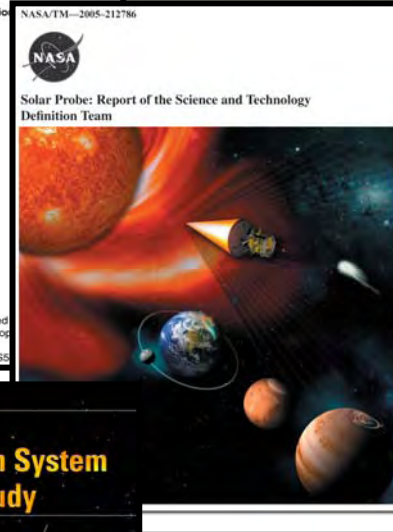
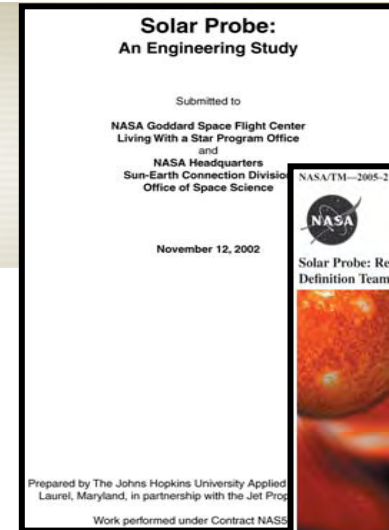
Envisioned S/W System Concept (Memory & Time protected apps, S/W Bus, Self-Healing)



LEON3FT Eval. Board
(Rad-Hard, SpW, High MIPS / Watts ratio)

Solar Probe

- 2002 Engineering Study
- 2005 Science & Technology Definition Team Report
- 2006 Thermal Protection System Risk Mitigation Study
- 2008 Solar Probe+ Mission Engineering Study



Solar Probe

was:

2005 STDT

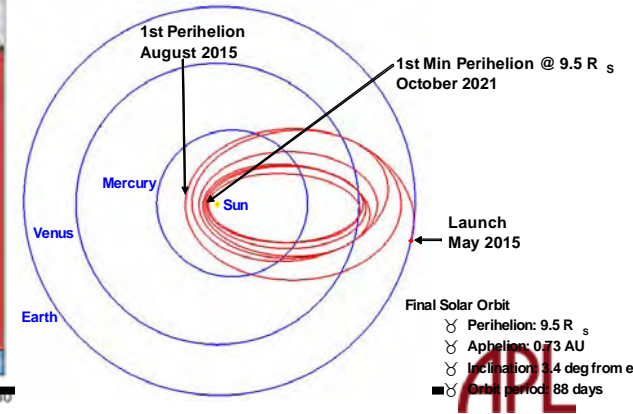
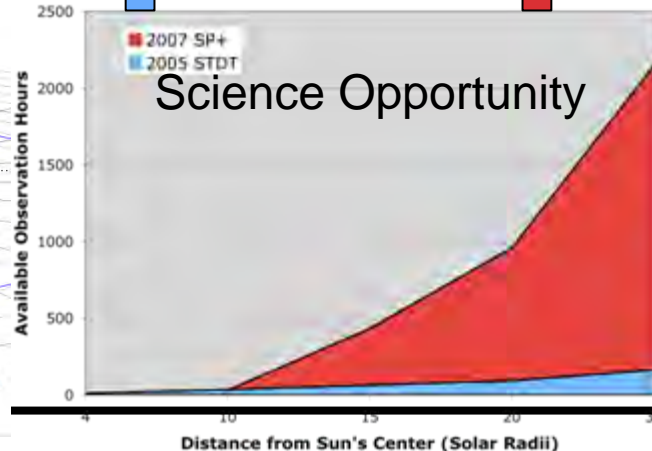
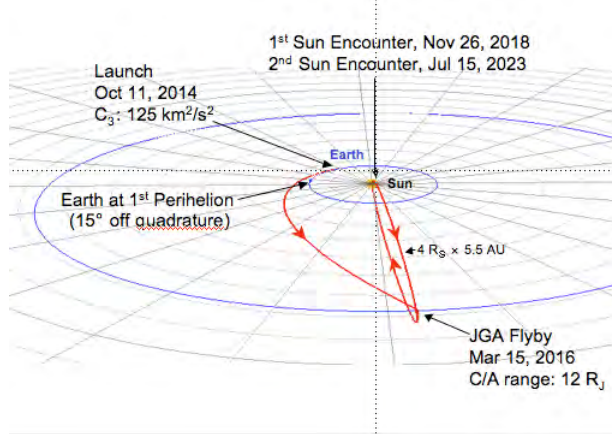
- \$1.2B (FY05 \$'s)
- Launch Oct 2014
- Atlas V 551, STAR-48, 672 kg (wet)
- Nuclear Power 375 W
- Closest Approach $4 R_{\text{Sun}}$ (0.02 AU)
 - 3000 Earth Suns
 - CC Heat Shield 2870°F
- Radiation 44 krad (100 mils AI)
- 2 Polar Orbits in 8.8 years
- Science (121 Gbits returned)
 - Solar Wind source, Energy flow, Acceleration mechanisms, Dusty Plasma phenomena
- 11 In-Situ & Remote Instruments



is now:

2007 SP+

- \$739M (FY07 \$'s)
- Launch May 2015(?)
- Atlas V 551, STAR-48BV, 481 kg (wet)
- Solar Power 482 W
- Closest Approach $9.5 R_{\text{Sun}}$ (0.044 AU)
 - 510 Earth Suns
 - CC Heat Shield 2600°F
- Radiation 30 krad (100 mils AI)
- 24 Equatorial Orbits in 6.9 years
- Science (2765 Gbits returned)
 - Goals same as SP 2005
 - Only the Polar Imager is deleted from the 2005 payload - 10 Instruments

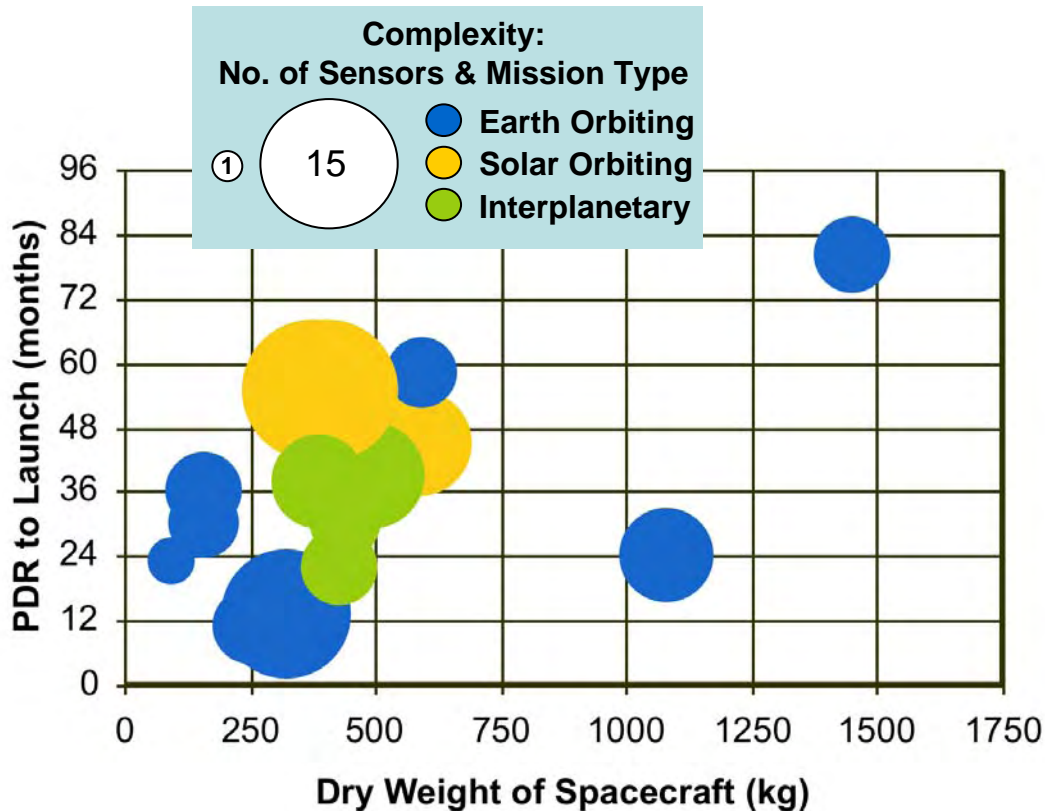


APL's Role in Exploration of the Solar System

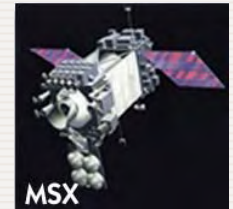
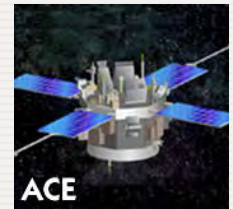
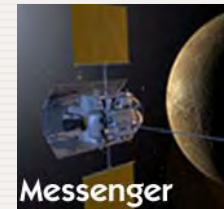
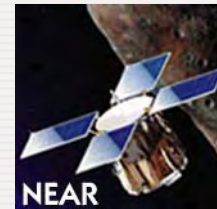


APL Track Record in Space

Innovative, Cost-Effective, End-to-End Space Missions



Recent Examples:



- 60+ Spacecraft
- 150+ Sensors & Payloads
- Short time to space
 - Modest-sized missions
 - Tight requirements process
 - Disciplined development



AS 9100 Registration

- JHU/APL is AS 9100 certified as of February 18, 2009.
- Scope of certification includes: Space and Earth science, analysis, research, technology development, program management, mission systems engineering, engineering, design, development, production, integration and test, and operation of spacecraft and instruments for both Civilian and National Security purposes.
- Continual monitoring of our Quality Management System will assure JHU/APL meets rigorous standards.



APL Space Mission Cost/Schedule History

Program	Sponsor	Launch Year	Bus Cost (FY05 \$M)	Weight of Bus (kg) Dry Mass	Cost Growth* (%)	Start to PDR (months)	PDR to Launch (months)	Schedule Growth** (months)	Mission Success
MAGSAT	NASA/GSFC	1979	30	158	8	3	30		Fully Successful
AMPTE	NASA/GSFC	1984	31	156	0	36			Fully Successful
GEOSAT-A	Navy	1985	38	543	7	3	37		Fully Successful
POLAR BEAR	USAF	1986	21	94	-5	5	23		Fully Successful
Delta 180	SDIO	1986	24	323	-3	3	13		Fully Successful
Delta 181	SDIO	1988	133	1081	-1	24			Fully Successful
Delta 183	SDIO	1989	25	253	8	3	11		Fully Successful
NEAR	NASA	1996	83	429	-7	30	22	0	Fully Successful
MSX	BMDO	1996	130	1449	13	10	38/80	0/42	Fully Successful
ACE	NASA/GSFC	1997	57	590	-10	30	45	0	Fully Successful
FUSE	NASA/GSFC	1999	22	264	3	43		0	Fully Successful
TIMED	NASA/GSFC	2001	85	591	10	30	37 / 58	0 / 21	Fully Successful
CONTOUR	NASA	2002	50	443	1	11	30	0	S/C lost during SRM burn
MESSENGER	NASA	2004	138	494	15	18	39	2 / 3	Successful To Date
New Horizons	NASA	2006	147	385	22	13	38	0	Successful To Date
STEREO	NASA/GSFC	2006	173	558/591	16	29	58	0/11	Successful To Date

Last updated 4/3/09

* Growth from cost estimate as of PDR

** # / #: First number shows slip attributable to APL; second number shows slip attributable to external factors (LV, scope change, etc.)



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