

# CONCEPT MATURITY LEVEL (CML)

## *Establishing A Shared Language To Articulate Maturity Of a Space-Science Mission Concept and Cost Uncertainty in the Early Formulation Phase*

**Presentation to the Planetary Science Decadal Survey  
Steering Group**

Keck Center  
Washington, DC

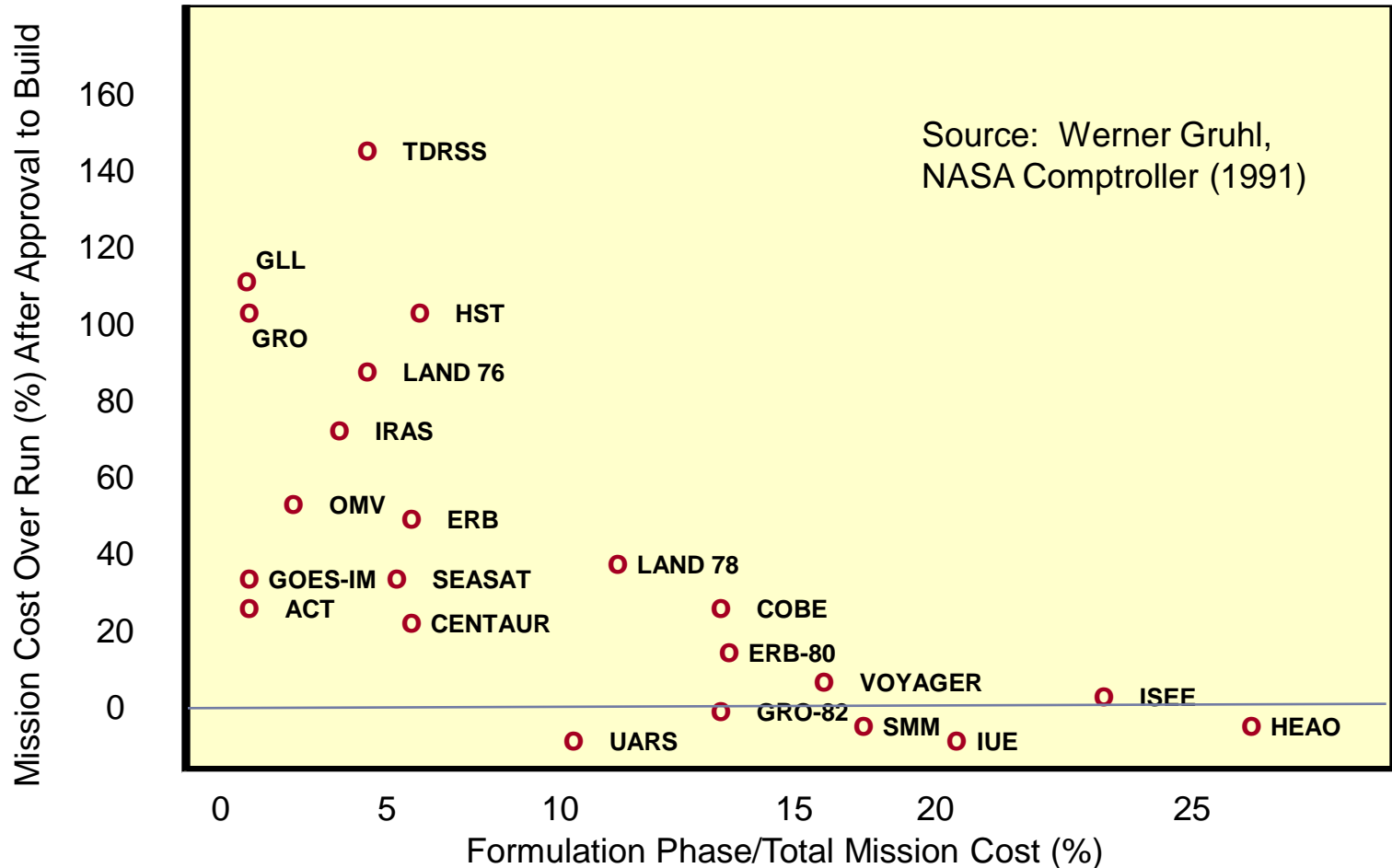
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# Key Challenge Facing the Planetary Science Decadal Survey (PS-DS)

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- Translate “An inventory of the top-level scientific questions” into (1) “the optimum balance among small, medium, and large missions,” (2) “a prioritized list of major flight investigations in the New Frontiers and larger classes,” and (3) “a list of important science goals which could be achieved by small spacecraft (Discovery and Scout class) missions”
  - To achieve objectives (1), (2) and (3) above, members of the Survey will have to create and sort through scores of mission concepts
  - A successful outcome **REQUIRES** that the concepts be at similar levels of maturity because ***cost uncertainty scales directly with concept maturity***
  - ***Hence, the Survey needs a tool for assessing the concept maturity level of every Decadal Survey mission concept***

# Mission Concept Maturity at “Approval” *vs* Subsequent Mission Implementation Cost Over Run

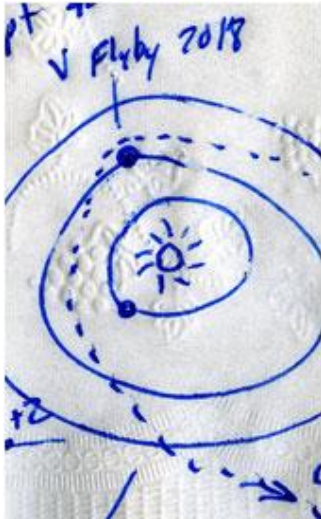


# Key Messages for the PS-DS, NRC and NASA

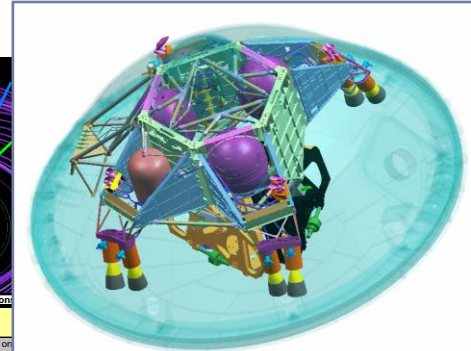
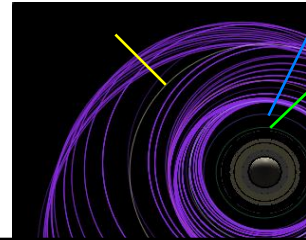
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1. An investment of 10-15% of total mission cost is required to reach 10-30% cost-estimate uncertainty
2. There is *no way* that the PS-DS or NASA can support such a level of investment for scores of mission concepts
3. Hence, NASA and the PS-DS must temper their expectations for what the Survey will produce
4. Focus instead on: Achieving a similar level of concept maturity and therefore cost uncertainty for ALL concepts to be consider.
  - **Concept Maturity Level (CML) is enabling**

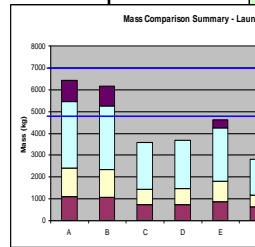
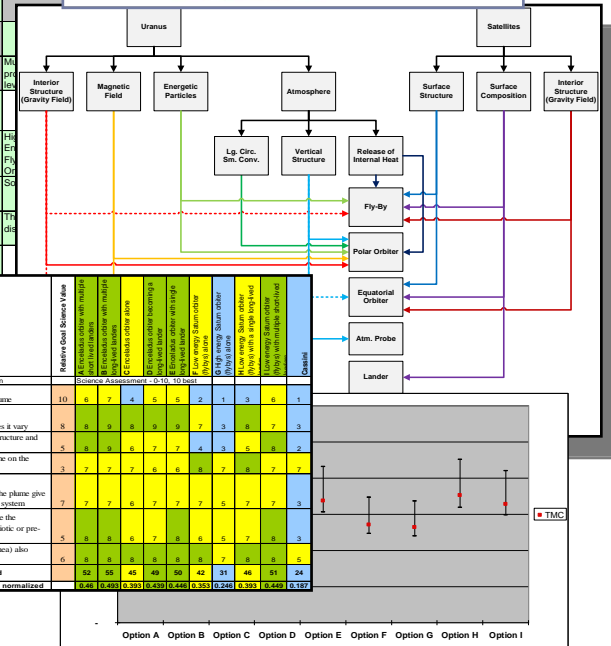
# Absent a Common Language and a Yardstick, One Person's Concept is Another's Doodle



or



Trades	Alternatives and Selection		
Launch vehicle	Atlas V	Delta IV-Heavy	Ares V
Cruise propulsion	SEP + GAs	Chemical + GAs	Propulsive on
Capture into Saturn system	Titan aerocapture (aerogravity assist)	Propulsive capture	
Pump-down mission design	Enceladus/Titan GAs only	Multiple moon GAs	
RPS type	MMRTG	ARPS (advanced Stirling)	
Orbiter implementation	Enceladus Orbiter	Low-Energy Enceladus Multiple-Flyby (Saturn Orbiter)	
Lander/Probe implementation	Fly-Through Probes and Impactors	Rough Landers	
Number of landers	None	One	
Lander lifetime/duration	Short-lived (~2 weeks on primary legions or fuel tank)	Long-lived (~1 year on RPS)	
Lander mobility type	SI		



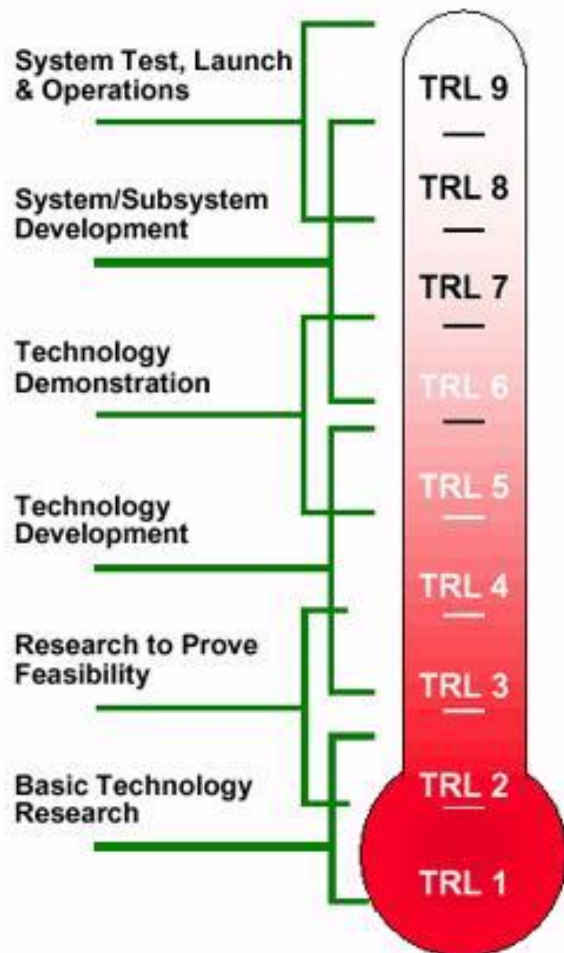
Science Goals, Enceladus Mission	Option A	Option B	Option C	Option D	Option E	Option F	Option G	Option H	Option I
1. What is the heat source, what drives the plume?	10	4	7	4	5	2	1	3	6
2. What is the plume production rate, and does it vary?	8	6	9	8	9	7	3	8	7
3. What are the effects of the plume on the structure and composition of Enceladus?	5	3	4	6	7	7	4	3	5
4. What are the interaction effects of the plume on the Saturnian system?	3	7	7	7	8	6	8	7	7
5. Does the composition and/or existence of the plume give us clues to the origin and evolution of the solar system?	7	7	7	8	7	7	5	7	7
6. Does the plume source environment provide the conditions necessary (or sufficient) to sustain biotic or pre-biotic chemistry?	5	8	8	7	8	6	5	7	8
7. Are there other bodies (Dione, Tethys, Rhea) also active, and if so, why not?	6	3	3	6	6	8	7	8	5
Value by Architecture, summed	52	55	45	49	59	42	31	46	51
Value by Architecture, weighted summed, normalized	0.66	0.43	0.33	0.43	0.46	0.33	0.26	0.33	0.44

**HIGH Cost Uncertainty**

**LOWER Cost Uncertainty**

# We Need a Language or a Yardstick – Just Like TRL... It Has Become a Powerful Communications Tool

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- ▶ TRL has become a universal language
  - ▶ Commonly used in AOs, briefings, conference sessions, peer-reviewed literature
  - ▶ NPR 7120.8 defines NASA-wide standard
  - ▶ “TRL 6 by PDR”
- ▶ ***TRL sets expectations despite variations in interpretation***

# Defining TRL Through Attributes

H/SW Both	Ques Catgry	% Complete	Attribute
			<b>TRL 6</b> (Check all that apply or use sliders)
B	T	78	Cross technology issue resolution and performance characteristic validations completed
H	M	100	Quality and reliability levels established
B	M	100	Frequent design changes occur
H	P	100	Draft design drawings are nearly complete
B	T	100	Operating environment for eventual system known
B	P	100	Collection of actual maintainability, reliability, and supportability data has been started
B	P	100	Design to cost goals identified
H	M	100	Investment needs for process and tooling determined
B	T	100	M&S used to simulate system performance in an operational environment
B	P	100	Final Test & Evaluation Master Plan (TEMP)
H	T	100	Factory acceptance testing of laboratory system in laboratory setting
B	T	100	Representative model / prototype tested in high-fidelity lab / simulated operational environment
B	T	100	Realistic environment outside the lab, but not the eventual operating environment
B	P	100	Final Systems Engineering Master Plan (SEMP)
S	T	100	Inventory of external interfaces completed
B	P	100	Technology Transition Agreement has been updated
B	P	100	Scaling issues that remain are identified and supporting analysis is complete
S	T	100	Analysis of timing constraints completed
S	T	100	Analysis of database structures and interfaces completed
B	P	100	Have begun to establish an interface control process
H	P	100	Draft production planning has been reviewed by end user and developer
H	M	100	Critical manufacturing processes prototyped
H	M	100	Most pre-production hardware is available
B	P	100	Technology Transition Agreement has been coordinated and approved by end user
S	T	100	Prototype implementation includes functionality to handle large scale realistic problems
S	T	100	Algorithms partially integrated with existing hardware / software systems
H	M	100	Materials, process, design, and integration methods have been employed
S	T	100	Individual modules tested to verify that the module components (functions) work together
B	P	100	Technology "system" specification complete
H	M	100	Components are functionally compatible with operational system
S	T	100	Representative software system or prototype demonstrated in a laboratory environment
B	T	100	Laboratory system is high-fidelity functional prototype of operational system
B	P	100	Formal configuration management program defined to control change process
B	M	100	Integration demonstrations have been completed
B	P	100	Final Technical Report
H	M	100	Production issues have been identified and major ones have been resolved
S	T	100	Limited software documentation available
S	P	100	Verification, Validation and Accreditation (VV&A) initiated
H	M	100	Process and tooling are mature
H	M	100	Production demonstrations are complete
S	P	100	"Alpha" version software has been released
B	T	100	Engineering feasibility fully demonstrated
B	P	100	Final Transition Plan with Business Case
B	P	100	Acquisition program milestones established
B	P	100	Value analysis includes business case
B	P	100	Technical alternatives include "do nothing case"
B	P	100	Formal requirements document available

▶ USAF scores 47 attributes for TRL 6

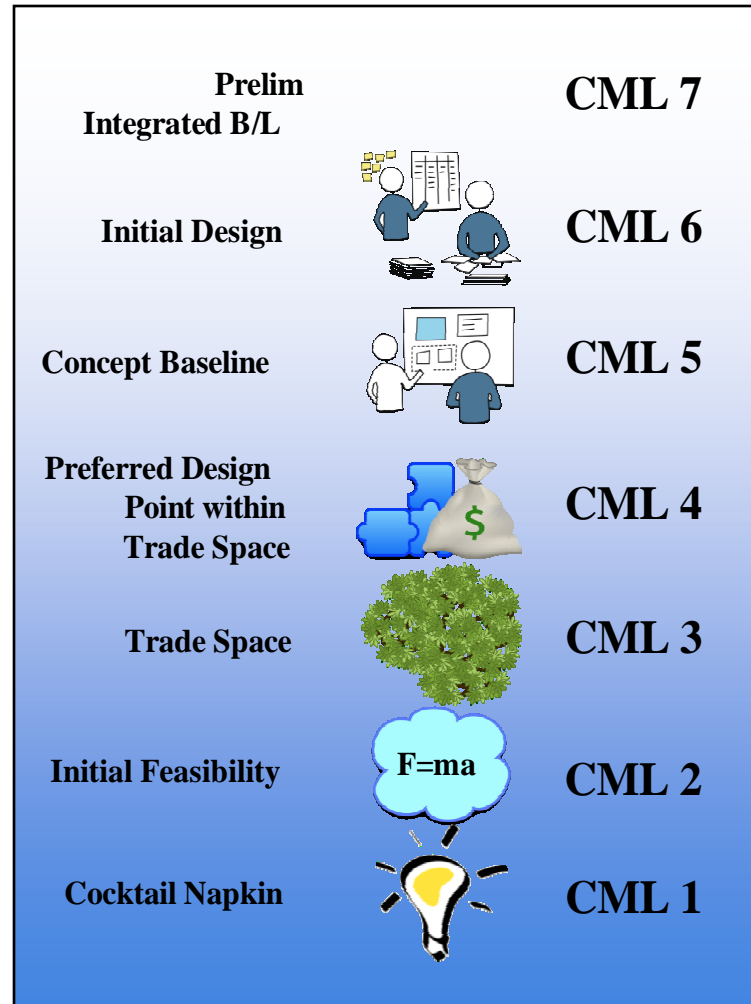
Draft design drawings are nearly complete

Draft production planning has been reviewed by end user and developer

"Alpha" version SW has been released

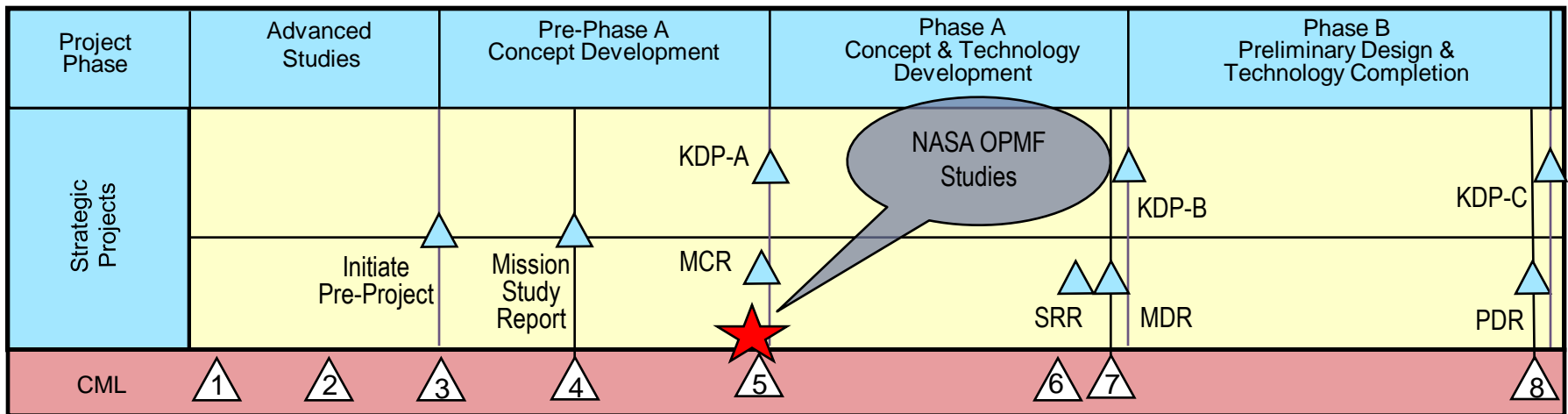
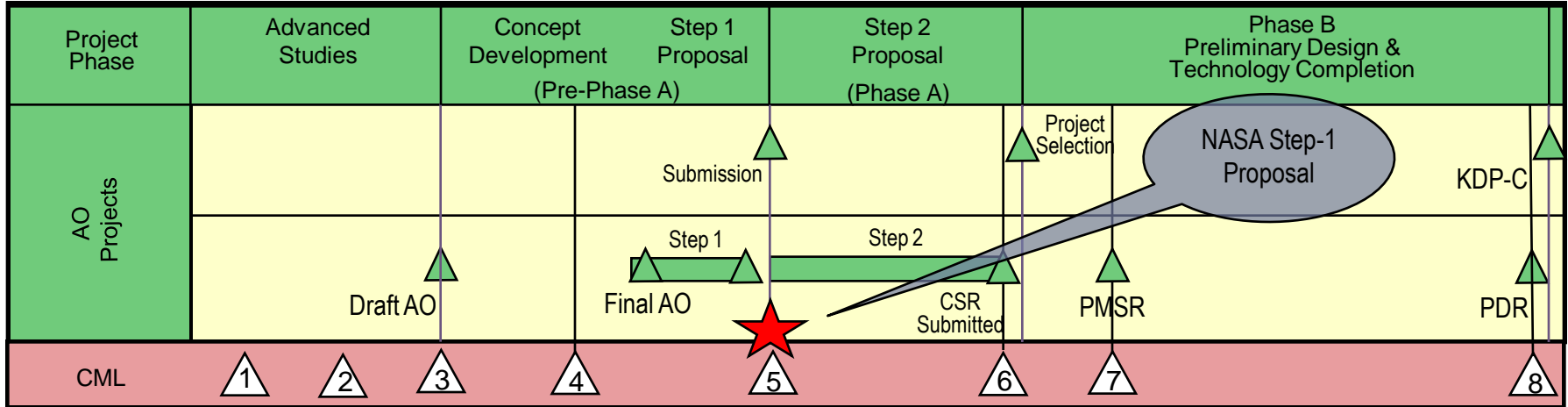
# Concept Maturity Level: CML

- ▶ Created by Mark Adler, JPL
- ▶ Inspired by TRL scale
- ▶ Covers the phase of mission concept formulation from “cocktail napkin” to NASA PDR
  - ▶ ***PDR = CML 8***
- ▶ Embraced by NASA PSD and now an integral part of JPL’s formulation approach





# CML and NASA Project Phase



# Attributes of a Mission Concept

## Technical Attributes (13)

- Requirements
- Mission Design
- S/C System Design
- Ground System Design
- Risk
- Technology
- Master Equipment List
- Technical Margins
- Explored Trade Space
- Mission Assurance Approach
- Modeling & Simulation Approach
- Launch Vehicle
- Planetary Protection

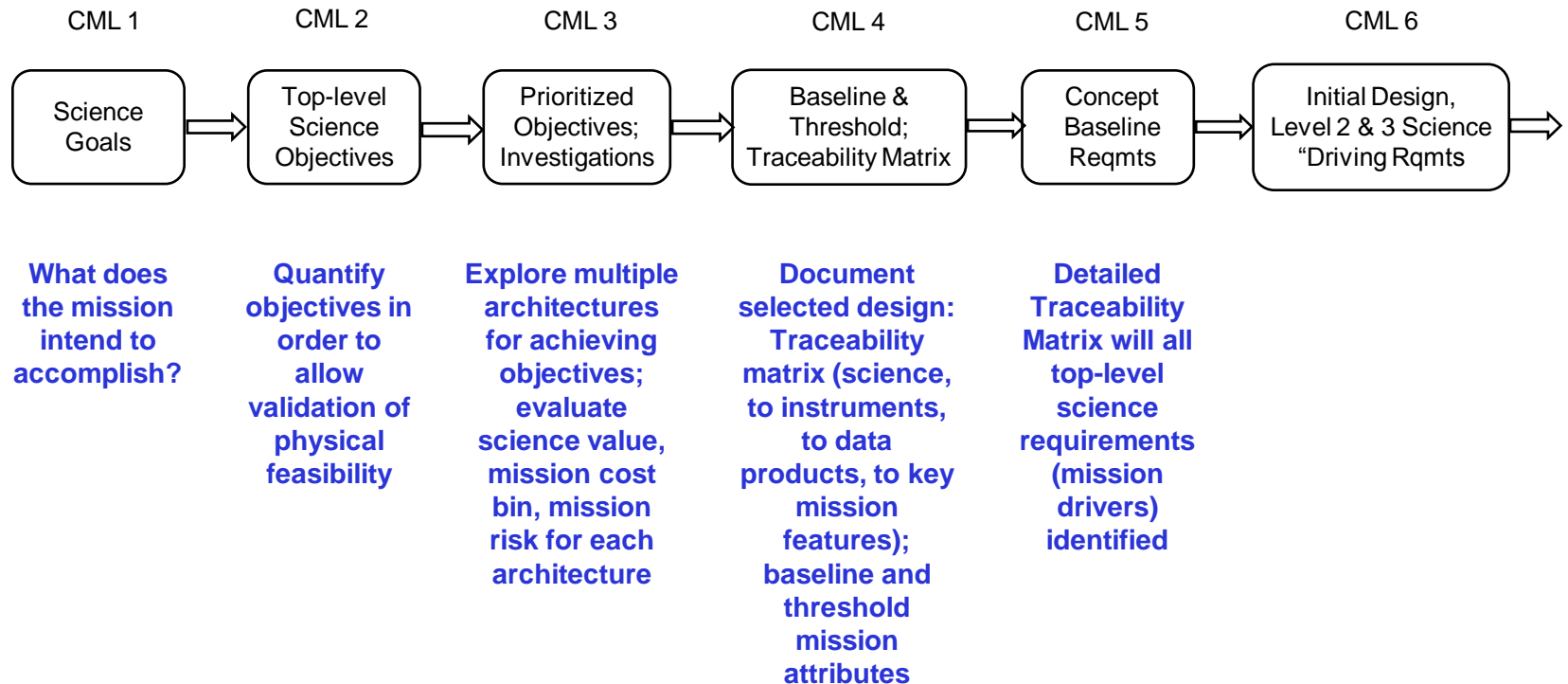
## Programmatic Attributes (10)

- Acquisition Approach and Make-Buy Decisions
- Project Organization
- Proposed Inheritance (value)
- Schedules
- Cost
- Documentation
- Work Breakdown Structure
- Test beds & Models
- Study Team Staffing
- Integrated Baseline

# Requirements: Increasing depth with increasing CML

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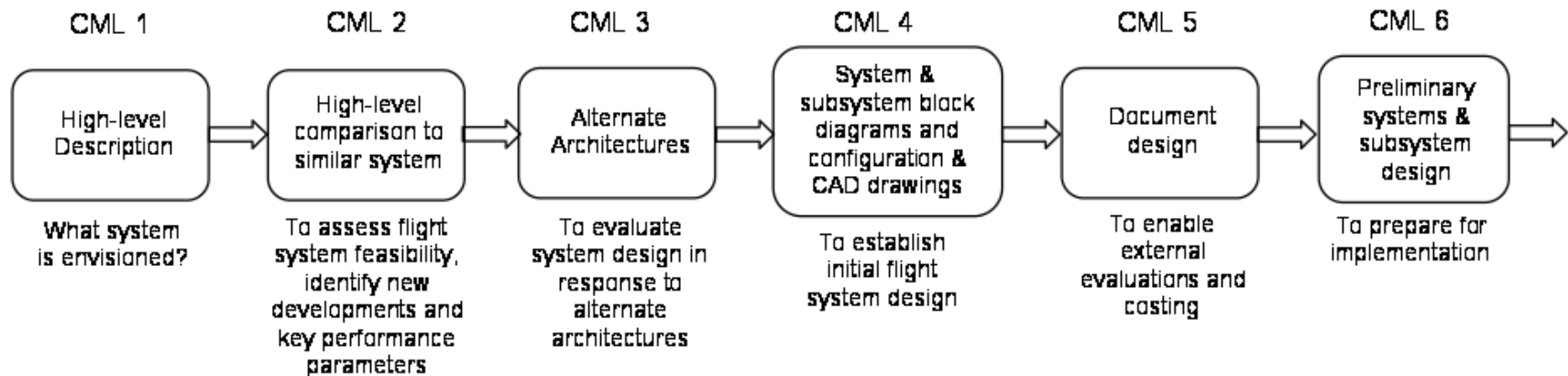
## 5.1.3 Science Objectives & Driving Requirements



# Spacecraft Design: Increasing depth and detail with increasing CML

## 5.5 Spacecraft & Instrument System Design

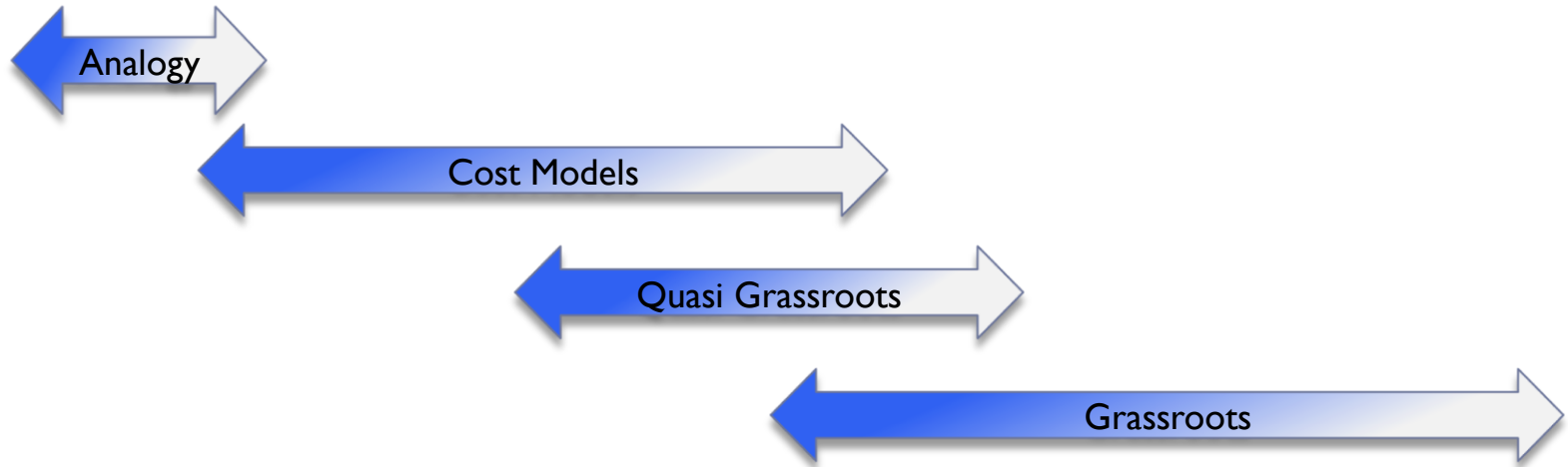
Spacecraft & instrument system design is based on Level 3 and 4 requirements. It is defined through a series of trade studies and continues to mature from a high level architecture to detailed design through the System Critical Design Review in Phase C.



# Appropriate Costing Methods at Each CML

Completed	Adv Studies	Adv Concept	Step 1 Proposal	Step 2 Proposal	Phase B		
Assigned	Adv Studies	Pre-Phase A		Phase A	Phase B		
CML	1	2	3	4	5	6	7

Costing Methodology



# Summary

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1. Cost uncertainty scales with mission concept maturity (and mission complexity)
2. Use of CML *alone* does not *directly* decrease cost uncertainty
3. It *does* enable a much more confident comparison and ranking of concepts within a given mission cost bin, by establishing that each concept is at a similar level of maturity, including the level of understanding of the mission complexity
4. ***Use of CML indirectly will decrease cost uncertainty by leading to more mature, better-understood mission concepts and...***
5. ***Markedly reduce the risk of prioritizing one immature, high-cost-uncertainty concept over another that is more mature and therefore has lower cost uncertainty.***